MISSION M108

Light Sport Aircraft General Specification





LAMBERT AIRCRAFT ENGINEERING

Mission M108 Description

Wings

Fabric covered wings supported by wing struts. Dual tubular aluminium alloy wing spars. Plywood wing ribs. Frise ailerons operated through cables. Manually operated trailing edge flaps. Quick wing folding system.

Fuselage

Fabric covered welded space frame fuselage structure from high strength steel alloy tubing. Horizontal tailplane supported by streamlined struts. Fixed fin with conventional rudder.

Landing gear

Fuselage mounted fixed landing gear. Main landing gear legs by cantilever composite leaf springs. 6" main wheels with hydraulic disc brakes. Different tyre size options. Steerable nose wheel with telescopic strut or steerable tailwheel.

Systems

Cable operated rudder. Elevator operated through push-pull rods. Aileron control through cable system. Elevator trim tab operated through push-pull cable. Composite fuel tanks in the wings leading to header tank in the fuselage. 14 VDC electrical system with battery and engine driven alternator.

Cockpit

Two seats in side-by-side arrangement. Fore and aft adjustable seats with backrest folding forward to provide access to baggage area. Full dual controls with elevator and ailerons operated by conventional stick columns. Differential toe brakes on tailwheel model; optional toe brakes on passenger side. Single lever operated brakes on nose-wheel model with optional parking brake.

Four point seat harnesses. High quality interior and seat upholstery. Central throttle control. Brake lever and elevator trim control between seats.

Two upward hinging doors. Cabin ventilation system. Optional cabin heating system.

Instruments and avionics

Choice of several standard avionics packages, each with various options. See list of avionics options.

Powerplant

Engine: Rotax 912iS Sport with DUC three blade ground adjustable propeller.



Mission M108 Equipment & Options

Landing gear and brakes	Nosewheel version	Taildragger version
Manual brake lever	●1	n/a
Parking brake valve	0	0
Toe brakes on pilot side	01	•
Toe brakes on passenger side	01	0
Main wheel tyres 15x6.00x6, nose wheel tyre 4.00x6	•	•
All tyres 8.00x6 (Carlisle)	0	0
Tyres 26"x12"x6" (Alaskan bush tyres)		o ²
Wheel spats	0	0

Exterior	All models
Exterior trim, standard trim scheme in 2 colours of choice	•
Aircraft exterior colour in other colour than white	0
Full custom paint scheme	0
Tinted windshield and windows (with anti-UV and IR coating)	0
Full transparent doors	0

Interior & Safety Equipment	All models
Four point seat harnesses (Schroth), TSO	•
Airframe emergency recovery system (ballistic parachute)	•
Seat cushions with energy absorbing foam	•
Stall warning system	•
Deluxe interior panels, floor carpets and seat upholstery (leather)	0
Cabin heating system	0
Heated pitot probe & alternate static source	0
LED Interior lighting system with dual dimmer	0
Whelen LED tail mounted anti-collision light (non-TSO)	0
Whelen wingtip mounted all LED anti-collision and position lights, TSO	0
Whelen LED leading edge landing/taxi/recognition light	0

Powerplant & Systems	Rotax 912iS
Rotax 912iS Sport engine, 73 kW (100 hp)	•
Thermostat controlled temperature of coolant & oil systems	•
40A main alternator (28A standard alternator retained as standby alternator)	0
DUC Swirl Inconel, 3-blade propeller, 68" diameter, metal leading edges	•
DUC Flash, 3-blade propeller, 68" diameter, metal leading edges	0
Long range fuel tanks (110 litres total usable fuel capacity)	0

 $^{^{\}rm 1}$ Nosewheel aircraft with optional toe brakes will not have a manual brake lever $^{\rm 2}$ Bush tyres will affect aircraft performance. Refer to POH supplement.

Standard

Mission M108 Specifications

Wing span 31 ft 4 in (9.57 m) Wing area 131 sq ft (12.2 m²)

Aspect ratio 7.45

Tailplane span 7 ft 11 in (2.40 m) Length 20 ft 0 in (6.10 m)

Height 7 ft 6 in (2.25 m) nosewheel 6 ft 2 in (1.90 m) taildragger

Wheel track 5 ft 11 in (1.77 m)
Width with wings folded 8 ft 6 in (2.60 m)
Baggage capacity 90 lbs (40 kg)

Baggage capacity 90 lbs (40 kg) Cabin width 43 in (1.08 m)

Fuel capacity, standard
Fuel capacity, optional

Never exceed speed

20.5 US gal (78 litres)
29.0 US gal (110 litres)
120 kt (222 km/h)

Max. crosswind comp. 12 kt (22 km/h) taildragger

16 kt (30 km/h) nosewheel

Limit load factors +3.8/-1.9 g

Mission M108 Performance

Engine Make Rotax

Type 912iS Sport

Number of cylinders 4

Displacement 82.6 cu in (1352 cm³)

Engine power 100 HP (73 kW) at 5800 rpm Approved fuels Mogas, Avgas 91UL & 100LL

Propeller Make DUC

Type Swirl Inconel

Number of blades 3

Diameter 68" (173 cm)

Empty weight 720-740 lbs (330-340 kg)

Maximum take-off weight 1320 lbs (600 kg)

Maximum level speed 105 kt (195 km/h)

Rate of climb 950 ft/min (4.75 m/s)

Cruising speed 85-90 kt (160-170 km/h)

Fuel consumption (typical) 12-15 litres/h (3.2-3.9 US gph)

Range, std fuel / opt. fuel 450 nm (850 km) / 680 nm (1250 km)

Stall speed, full flaps 38 kt (70 km/h)

Take-off ground roll 250-400 ft (80-130 m)

